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**DGAC AIRWORTHINESS DIRECTIVE 88-055
STAMPE SERIES AIRCRAFT
REAR BRACKET - FITTING OF WING CENTRAL UNIT**

This letter transmits a copy of DGAC Airworthiness Directive 88-055 in respect of Stampe Series aircraft.

In accordance with Airworthiness Notice No. 36 this Airworthiness Directive is mandatory for applicable aircraft on the UK Register.

A handwritten signature in black ink, appearing to read 'R J Teew'.

R J TEW
Maintenance Approvals Section

CABANT REAR STEETS
New Bolts Used (All positions)
New SS14 fittings

AIRWORTHINESS DIRECTIVE
issued by the
Direction Generale de l'Aviation Civile

STAMPE Aircraft
Rear bracket-fitting of wing central unit

The present Airworthiness Directive concerns STAMPE aircraft types SV4 A/ SV 4 A 1 / SV 4 B / SV 4 C / SV 4C 1 / SV 4L 150.

Some of the fixing bolts of the two rear port and starboard bracket-fittings supporting the rear struts of the top wing central unit have been found to be very badly corroded (to the extent of causing them to break around the threading).

Each bracket is fixed to the top spar of the fuselage by 4 horizontal and 2 vertical screwbolts, accessible from the post in front. They are locked by centre punching on the external sides of the post.

1. Before any further flight is undertaken following the date of enforcement of this Airworthiness Directive, check the grip of the horizontal and vertical screwbolts by means of a screwdriver.
 - 1.1. If a screw turns without it being possible to tighten it (it has probably sheared), withdraw the aircraft from service and remove this screw by applying AEROSPATIALE STAMPE Service Bulletin no. 4.
 - 1.2. If the tightness check is satisfactory, check or repeat the locking with a centre punch to the screw-head side.
2. Within 4 months from the date of enforcement of this present Airworthiness Directive, unless this examination has been carried out less than 4 years previously, remove the screws individually in the following manner: unscrew, withdraw, inspect, then if the condition is satisfactory, rescrew and tighten before dealing with another screw in the same manner, so as to avoid upsetting the tension of the bracing wires. Relock after completing the operation by using blue LOCTITE.

The faulty screws are to be replaced by applying the Constructor's instructions which are the subject of Chapter IV of AEROSPATIALE STAMPE Service Bulletin no. 4.

3. Amend the maintenance programme as follows:
 - 3.1. Include, in the operations to be done every 100 flying hours, the inspection which is the subject of paragraph 1 of this Directive.
 - 3.2. Include, in the operations to be done during the detailed inspection every 4 years, the inspection work which is the subject of paragraph 2 of this Directive.
4. Record the execution of the inspections and the work in the aircraft's log.

cf: AEROSPATIALE STAMPE Service Bulletin no. 4

DATE OF ENFORCEMENT: 23RD APRIL 1988